

# Introduction to the Design of Space Mechanisms

Theme 2:  
Environmental  
constraints  
Part 2



Gilles Feusier

# Space Environment Constraints

## Main environmental constraints:

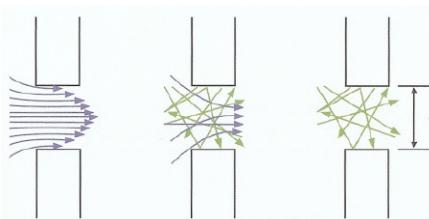
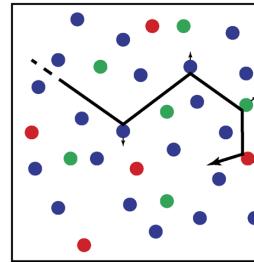
## ▪ Vacuum

- Mean free path  $\lambda$ , molecular flow
  - Heat exchange: no convection
  - Outgassing of a cavity
  - Evaporation of materials
  - Breakdown Voltage

## ▪ **Radiations**

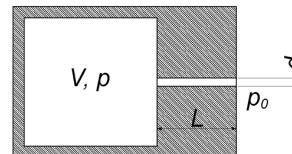
- Electromagnetic and particles
  - ATOX: Atomic Oxygen (O, UV's)

## ▪ **Vibrations and Shocks**



$$K_n < 0.01$$

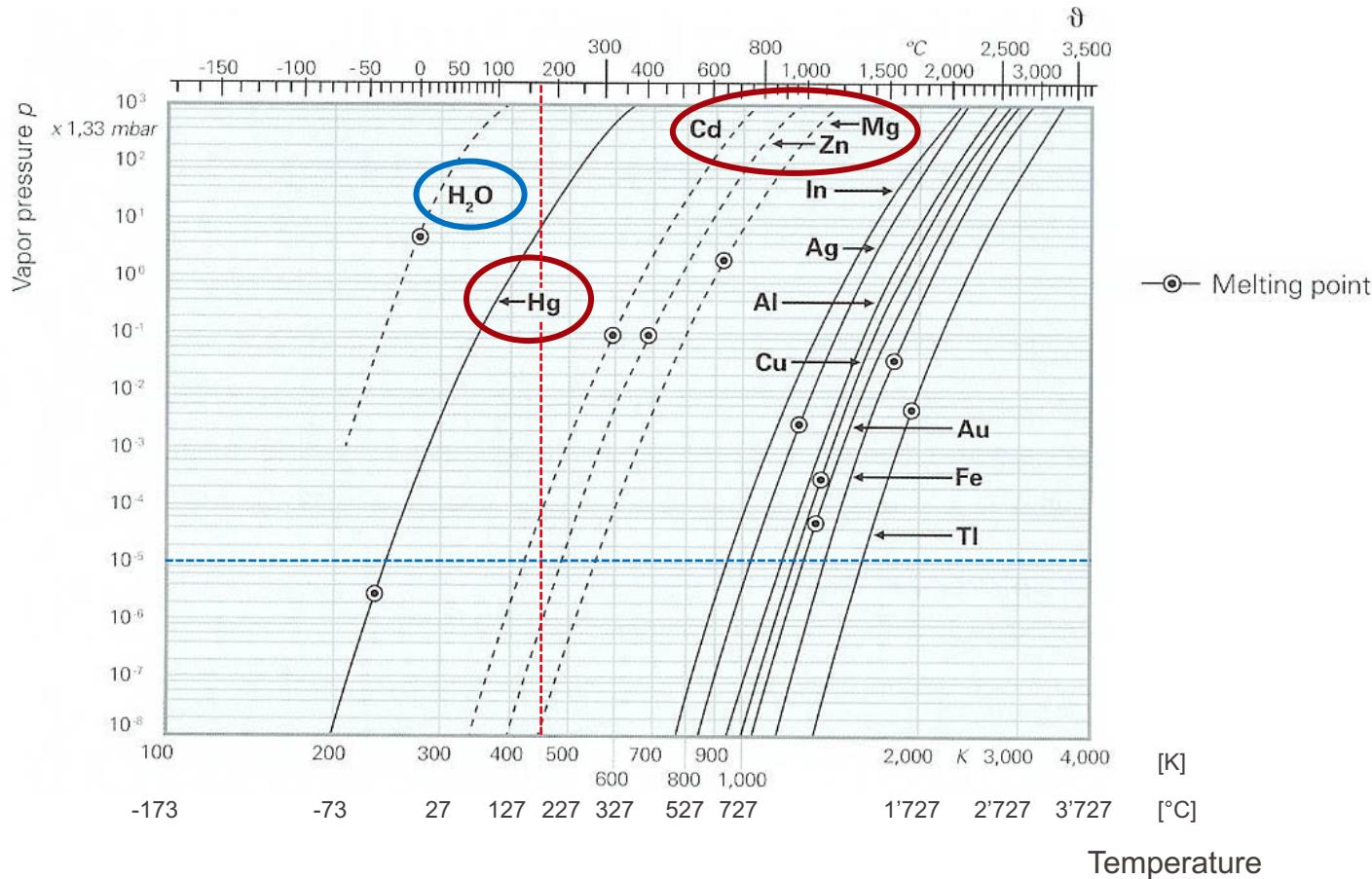
$$K_n > 0.5$$



### Conductance ( $C$ )

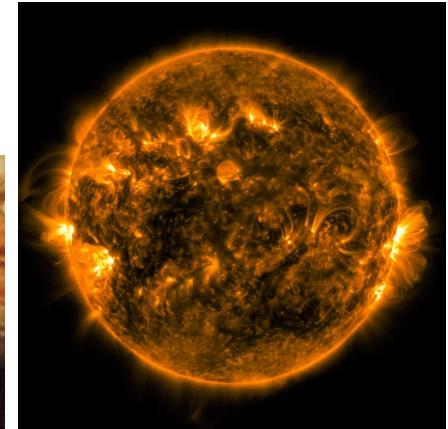
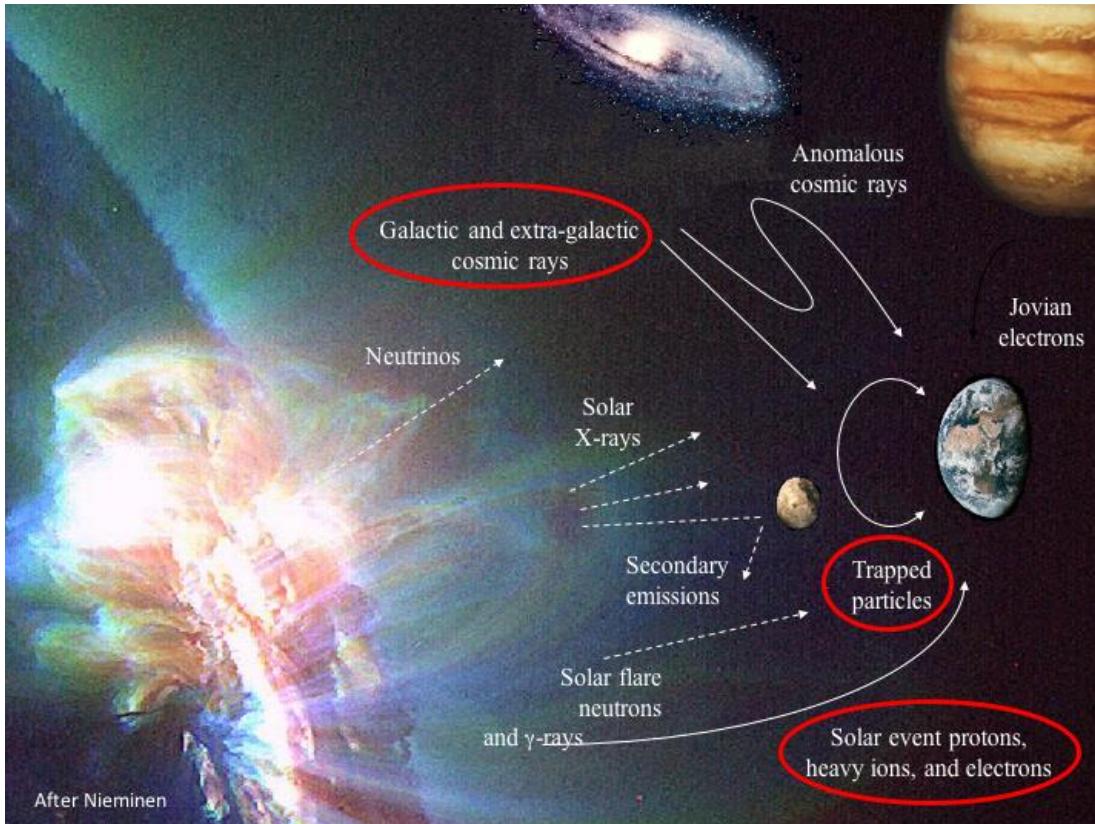
$$Q = C \cdot (p - p_0)$$

# Vapor Pressure



# Radiations

- Source of radiation in space



Source: NASA/SDO

**Versoix, Switzerland, May 10<sup>th</sup>, 2024**  
G5 (Extreme) acc. to NOAA



Source: Lionel Peyraud / Meteosuisse

Source: C.Poivey, Radiation Effects in Space Electronics (ESA 2019),

- Two types of radiation are particularly dangerous for humans:
  - **High energy Galactic Cosmic Rays (GCR)**
    - Various types of particles
  - Solar radiations: **Solar Particle Events (SPE)**
    - Mainly protons, lower energy than GCR, but much higher quantity

Reading:  
[2.3]

**COMPUTER SIMULATIONS OF RADIATION SHIELDING  
IN SPACE BY POLYMERIC MATERIALS**

Christopher A. O'Neill

College of William and Mary, Virginia (2006)

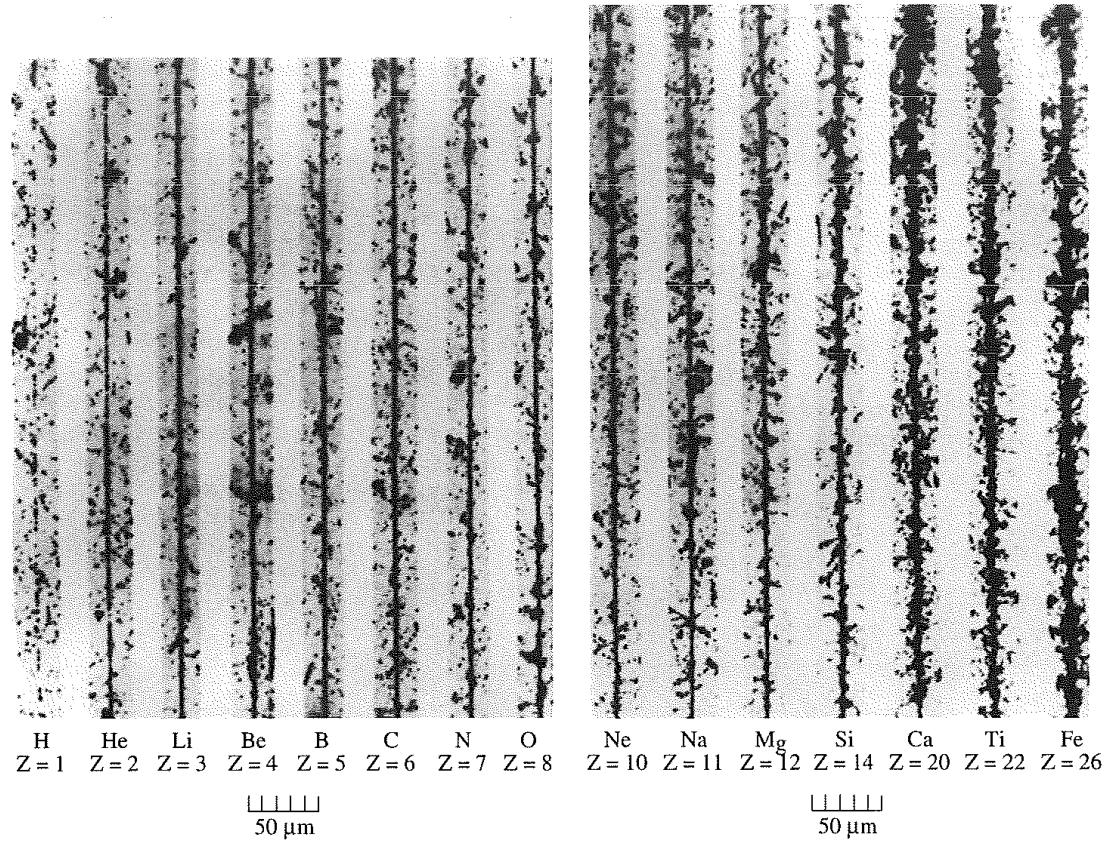
Advisor: Dr. Robert A. Orwoll, Ph.D

## Cosmic-ray ion tracks in nuclear emulsion

Effect of the interaction of charged particles with the matter.

Different Z materials,  
same energy of the  
incident beam.

Source: J. W. Wilson et al., NASA Conference  
Publication 3360 (1997) p.13 / Taken from McDonald,  
NASA TM X-55245 (1965)



# Ionizing Radiations: best absorbing materials

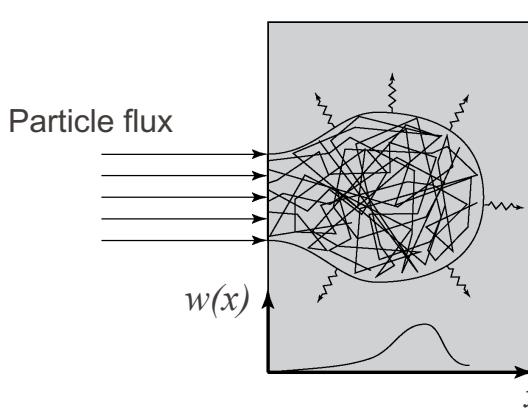
- most effective radiation shielding materials (i.e. producing less secondary electrons),
  - have the highest electron density,
  - the least electronic excitation energy,
  - the least tight binding corrections for the inner shell electrons

→ Hydrogen

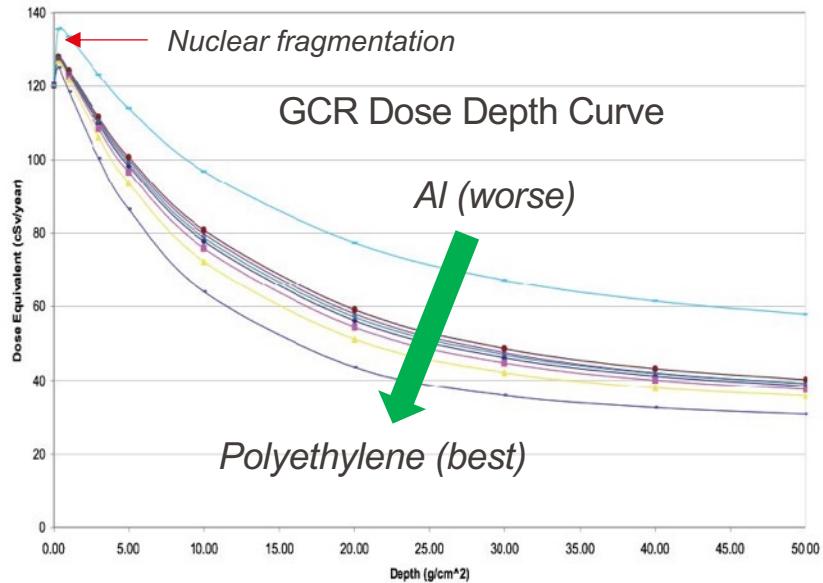
Shielding equivalence in water column heights:

Atmosphere:	10 m
Typical spacecraft:	20 cm
Space suit:	1.5 cm

Source: T. Berger, "Radiation dosimetry onboard the International Space Station ISS",  
Z. Med. Phys. 18 (2008) 265–275



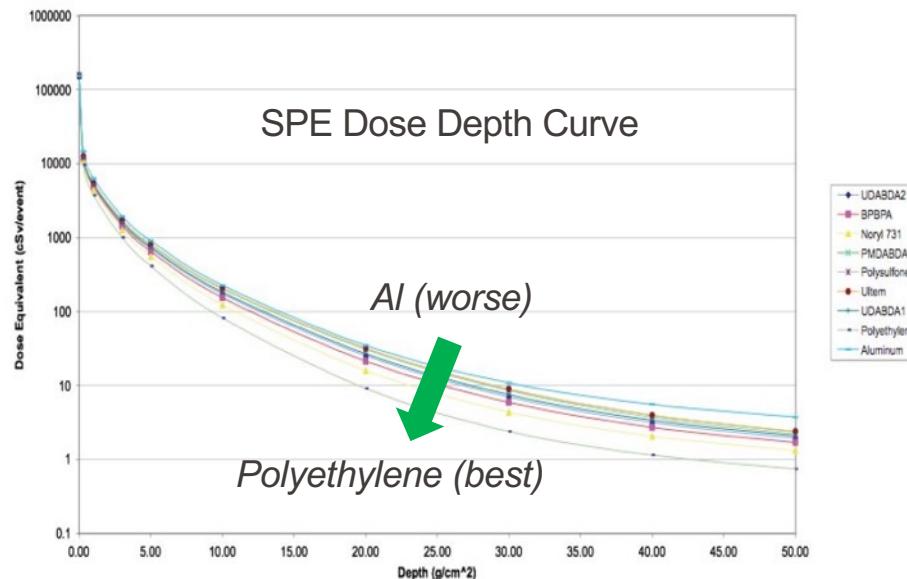
# Ionizing Radiations: Manned Space Flight



GCR: Galactic Cosmic Rays  
 SPE: Solar Particle Events

Source: Christopher A. O'Neill "Computer Simulations of Radiation Shielding in Space by Polymeric Materials"

## Effect of shielding



# Ionizing Radiations: Manned Space Flight

- Radiation flux, fluence:  $\phi = \frac{dN}{dA}$  [m<sup>-2</sup>]

Where where  $dN$  is the number of particles incident on a sphere of cross-sectional area  $dA$

- Energy fluence:  $\psi = \frac{dR}{dA}$  [J/m<sup>2</sup>] or [eV/m<sup>2</sup>]

Where where  $dR$  is the radiant energy incident on a sphere of cross-sectional area  $dA$

- Dose:  $D = \frac{d\bar{\varepsilon}}{dm}$  [J/kg = Gy (Gray)]

Mean energy  $\bar{\varepsilon}$  absorbed by unit mass  $m$

- Equivalent dose:  $H_T = \sum w_R \cdot D_{T,R}$  [Sv (Sievert) = J/kg]

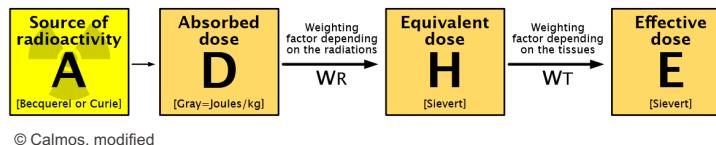
- Effective dose:  $E = \sum w_T \cdot \sum w_R \cdot D_{T,R}$  [Sv]

Dose in a tissue ( $D_{T,R}$ ) weighted by a factor depending of the type of radiation and its energy ( $w_R$ ) and a factor depending of the physiological effect on the tissue ( $w_T$ ).

# Ionizing Radiations: Manned Space Flight

Type and energy range	Radiation weighting factor, $w_R$	
Photons, all energies	1	
Electrons and muons, all energies	1	
Neutrons, energy	<10 keV	5
	10 keV to 100 keV	10
	100 keV to 2 MeV	20
	2 MeV to 20 MeV	10
	>20 MeV	5
Protons, other than recoil protons, energy >2 MeV	5	
Alpha particles, fission fragments, heavy nuclei	20	

Source: ECSS-E-ST-10-12C [2.2]



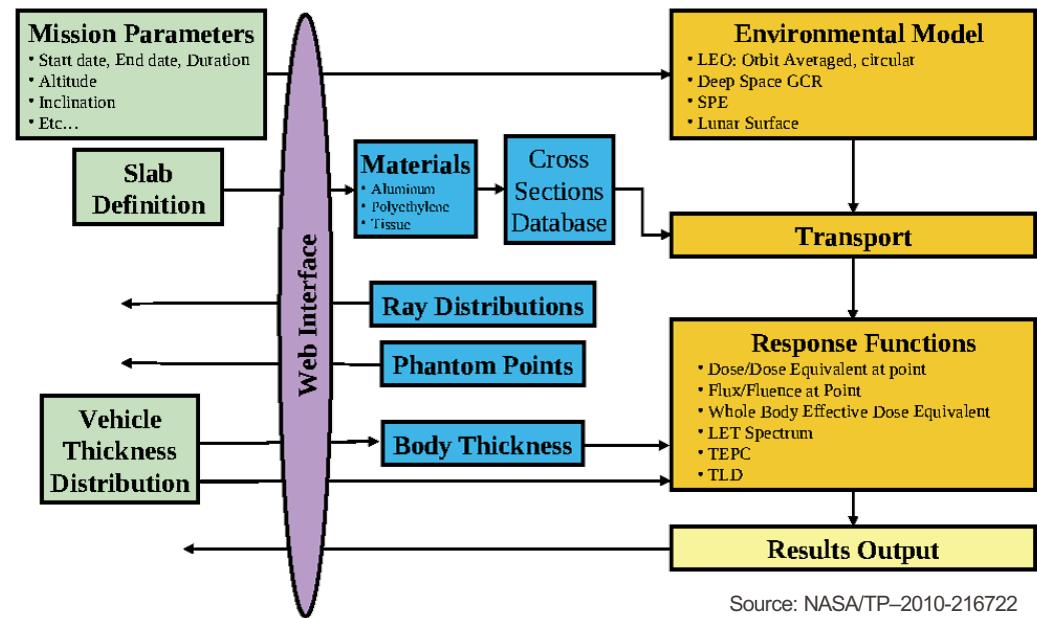
## Exercise 2.2: Radiations

Organ or tissue	Tissue weighting factor, $w_T$
Gonads	0,20
Bone marrow (red)	0,12
Colon	0,12
Lung	0,12
Stomach	0,12
Bladder	0,05
Breast	0,05
Liver	0,05
Oesophagus	0,05
Thyroid	0,05
Skin	0,01
Bone surface	0,01
Other tissues and organs	0,05

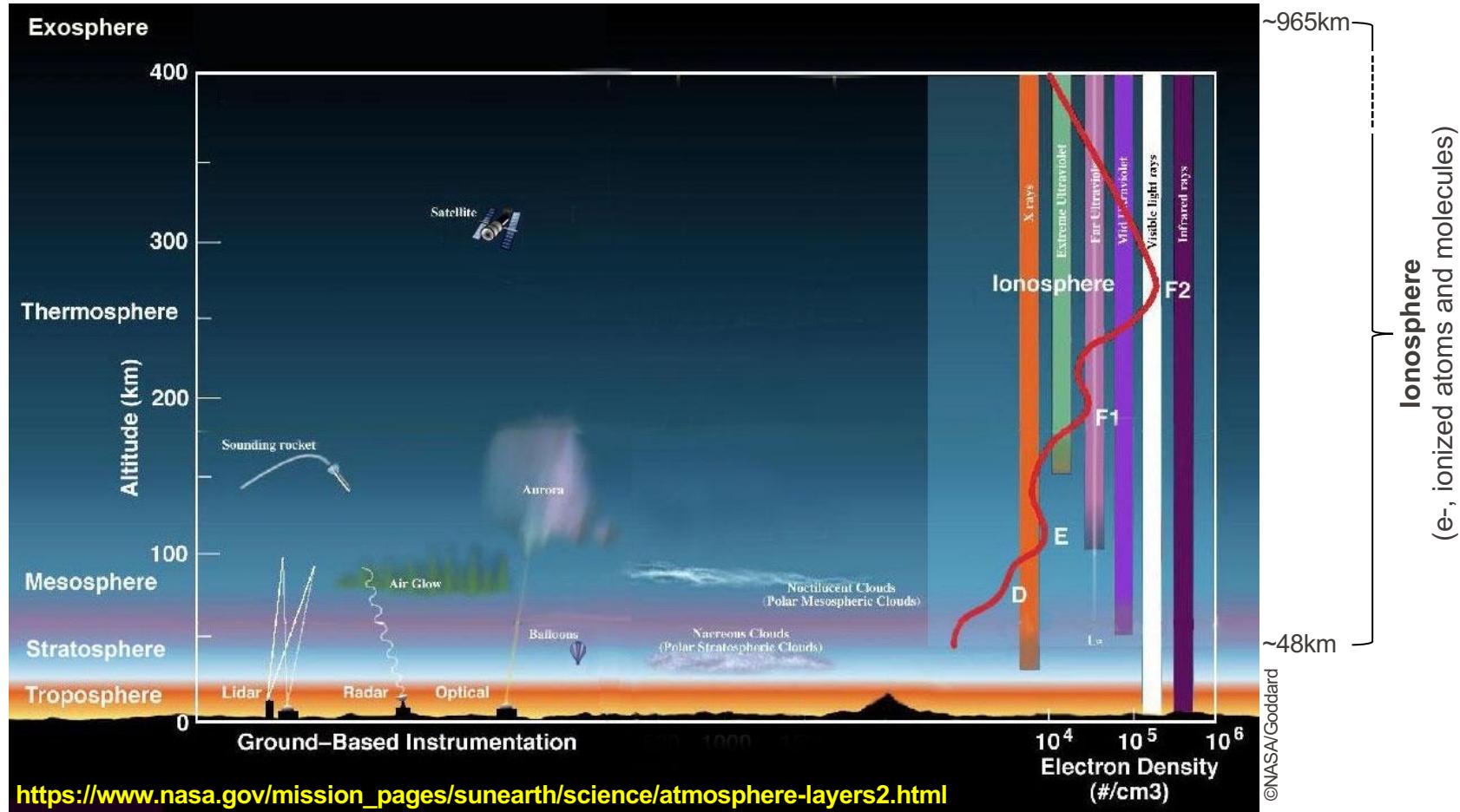
# Ionizing Radiations: Tools

- ECSS-E-HB-10-12A - Calculation of radiation and its effects and margin policy handbook [2.7]
- ECSS-E-ST-10-12C - Methods for the calculation of radiation received and its effects, and a policy for design margins [2.2]
- ...

- OLTARIS (On-Line Tool for the Assessment of Radiation in Space):  
<https://oltaris.larc.nasa.gov>



# Ionizing Radiations: Absorption by atmosphere



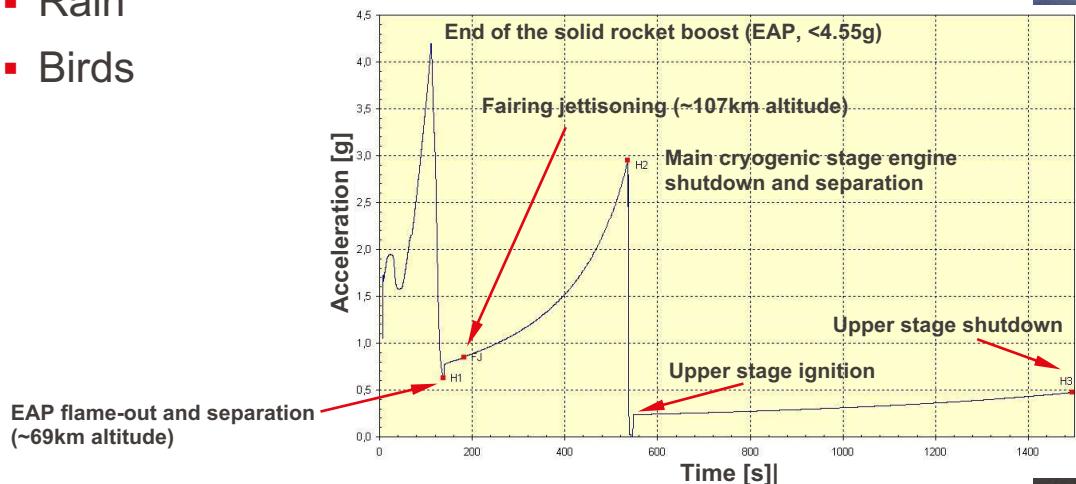
# Launch and Ascent

- Vibrations (sine, random, acoustic)
- Accelerations
- Shocks
- Thermal Flux
- Lightening impact
- Rain
- Birds

Source: NASA



Longitudinal static acceleration

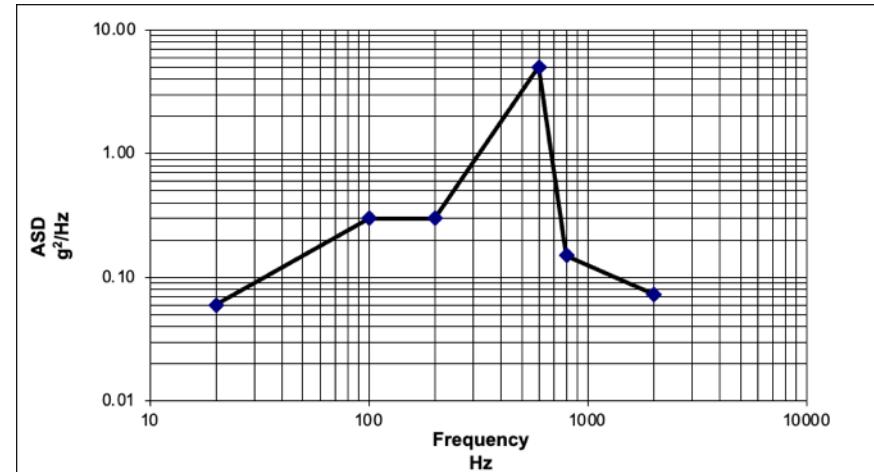


Source: Ariane 5 User's Manual Issue 5 Revision 2

- The vibration loads injected into a mechanism during the launch to orbit depend on:
  - the launcher (Ariane 5, Falcon 9, ...)
  - the attachment point of the mechanism
- Typical vibration load levels for a mechanism mounted on a spacecraft launched by an Ariane 5 rocket (case of the ATV)
  - Overall Grms: 35 g<sub>rms</sub>

Frequency [Hz]	ASD [g <sup>2</sup> /Hz]	Slope [dB/Oct]
20 to 100	0.06 to 0.30	3.01
100 to 200	0.30	0.00
200 to 600	0.30 to 5.00	7.71
600 to 800	5.00 to 0.15	-36.69
800 to 2000	0.15 to 0.073	-2.37

■ ASD: Acceleration Spectral Density



- Given: Acceleration Spectral Density (ASD) in  $\text{g}^2/\text{Hz}$

*Note: We should speak of Acceleration Spectral Density (ASD) for the injected acceleration.  
Power Spectral Density (PSD) refers formally to the measured plot (values of the accelerometers).*

- Root-Mean-Square Acceleration

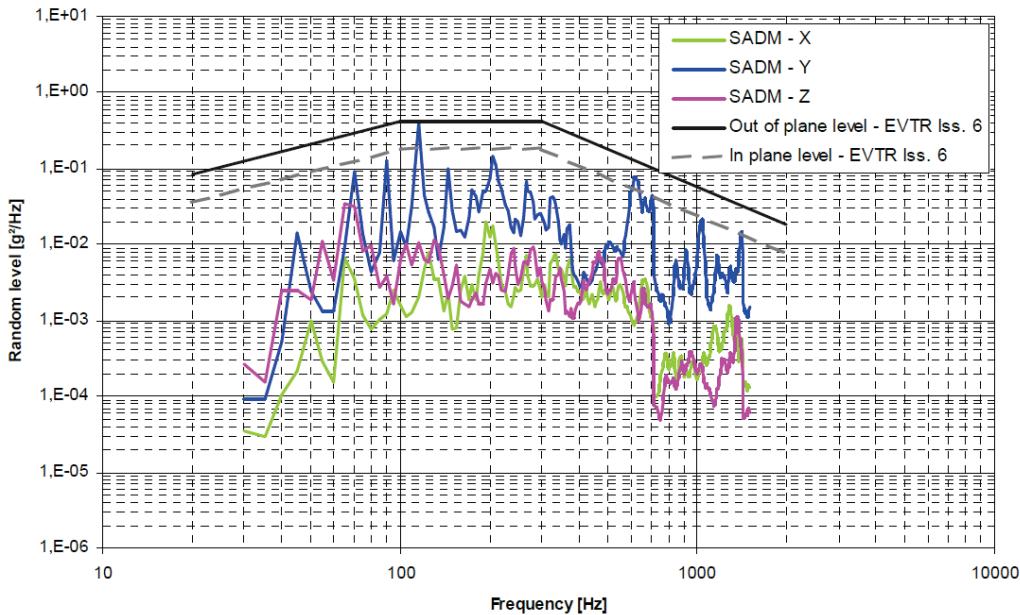
$$\bar{A} = \sqrt{\int_{v_L}^{v_H} \text{ASD}(v) \cdot dv} \quad [\text{g}_{\text{rms}}]$$

- It is the average acceleration load injected into the mechanism.
- The mechanism may have eigenfrequencies leading to much higher effective amplitude for some of its components.
- The concept of “overstress” will be introduced later on.

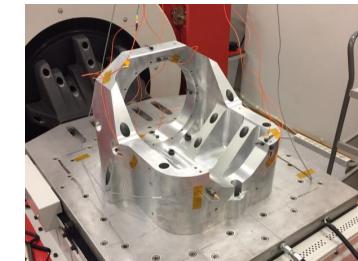
- *Reminder: an octave is a doubling of  $v$ :*  $\# \text{Octave} = \frac{\log(v_H/v_L)}{\log(2)}$

# Random Vibrations

- Vibroacoustic analysis of random vibration levels for Sentinel 3 SADM



Source: ECSS-E-HB-32-26A Space engineering -  
Spacecraft mechanical loads analysis handbook [2.5]



Source: Elkjær et al. ESMATS 2019



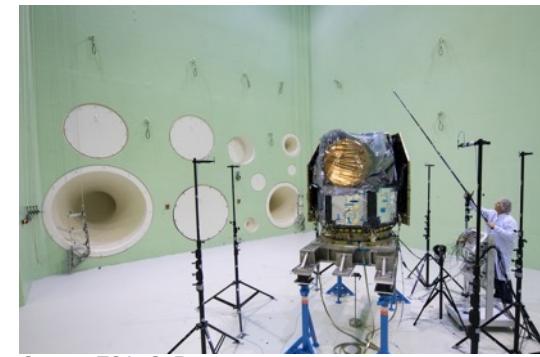
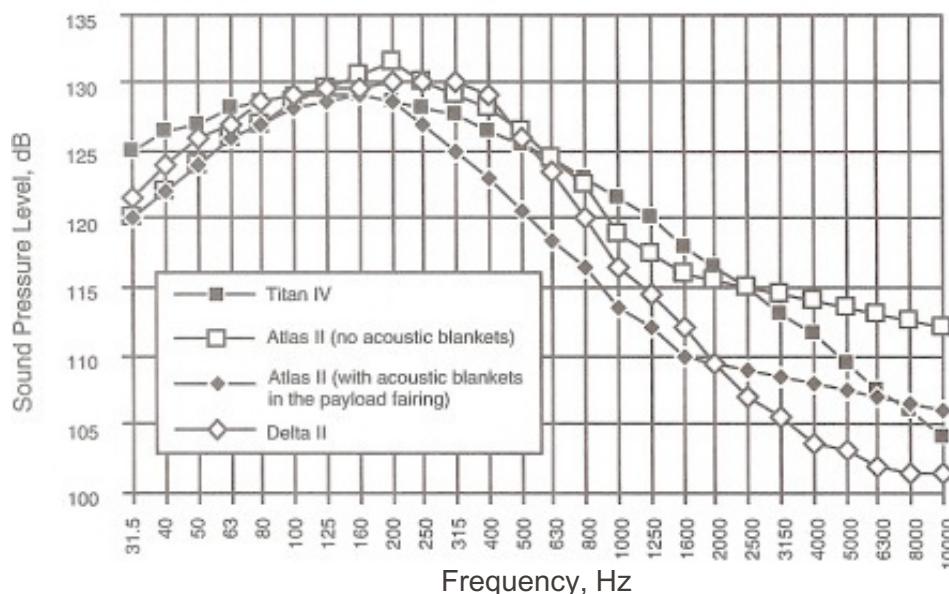
Source: Spanoudakis et al.,  
ESMATS 2019

# Launch Acoustic Noise

- $SPL(\nu)$ : Sound Pressure Level as a function of the frequency  $\nu$ .

$$SPL(\nu) = 20 \cdot \log_{10} \frac{P(\nu)}{P_{ref}} \quad [\text{dB}]$$

$$P_{ref} = 2 \cdot 10^{-5} \text{ Pa} \quad (\text{threshold of hearing})$$

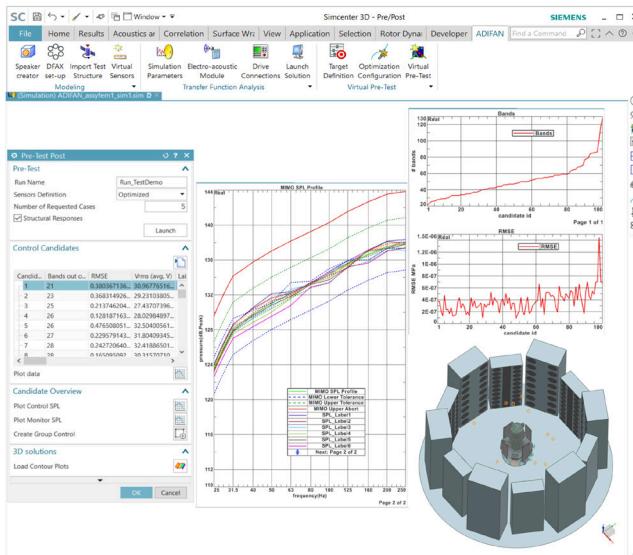
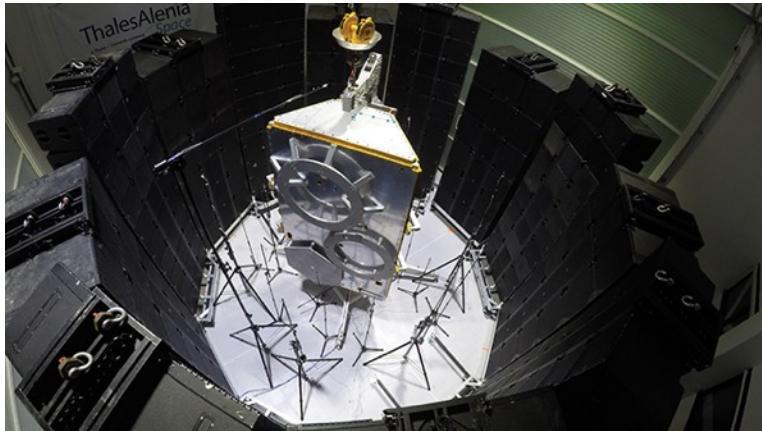
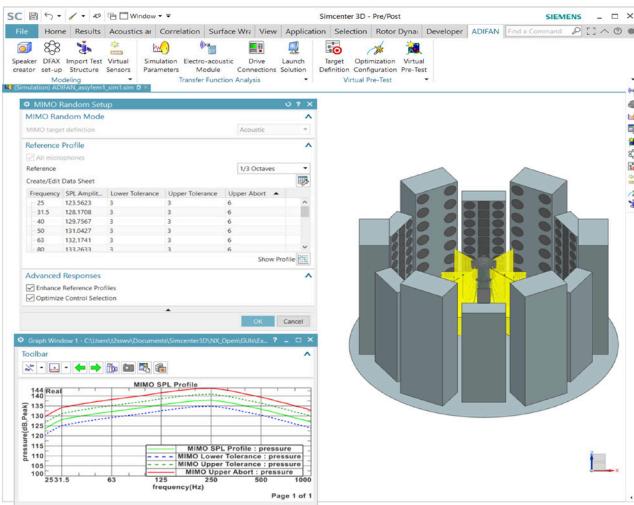


Source: ESA-G. Porter

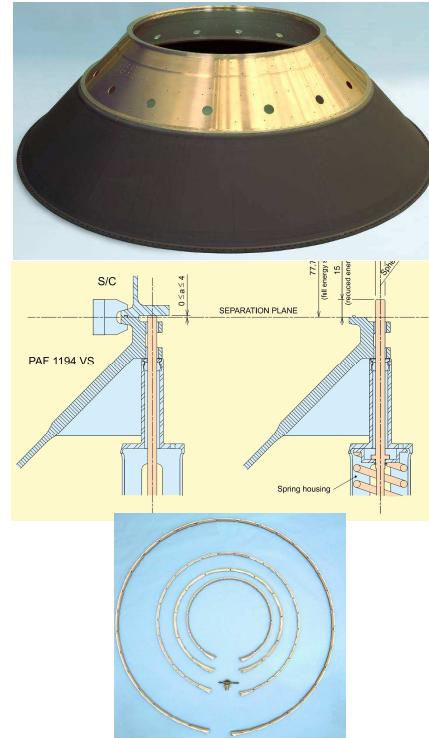
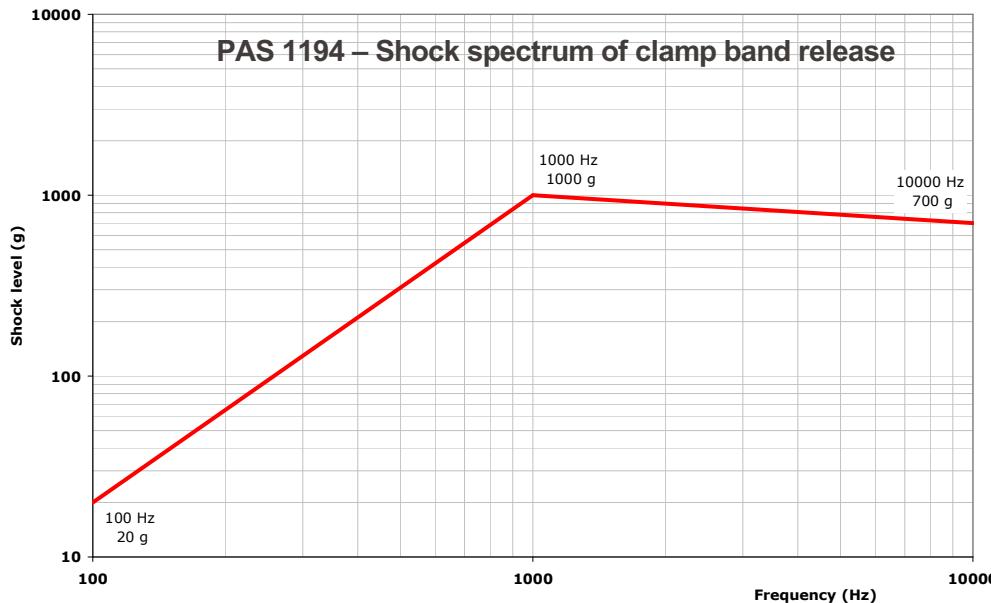
Source: Th. P. Sarafin (ed.),  
Spacecraft, Structures and  
Mechanisms, Wiley J. Larson,  
Managing ed., 2003, fig.3.7, p. 45 [2.4]

# Launch Acoustic Noise

- Digital Twins
  - First: NASA 2010
  - Industry 4.0
  - Artificial Intelligence, Machine Learning, IOT ...



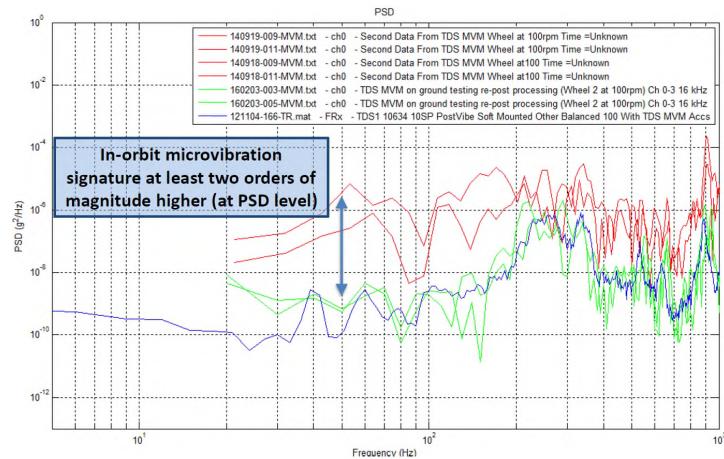
- Main events producing noticeable shocks:
  - the launch vehicle upper stage separation from the main cryogenic stage
  - the fairing jettisoning
  - the spacecraft separation
  - sub-system release and deployment



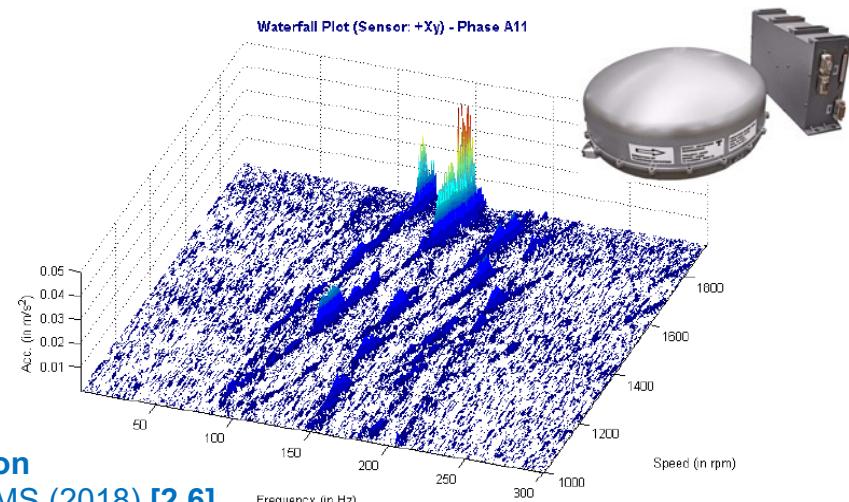
Source: Ariane 5 User's Manual Issue 5 Revision 2

# Micro-vibrations

- Has effects on:
  - Micro-g space laboratory experiments
  - Attitude control of spacecraft
  - Quality of images for Earth Observation (EO) spacecrafts or space telescopes
  - ...
- Can be produced by operating mechanisms and human induced vibrations:
  - Reaction wheels, momentum wheels,
  - Pumps, valves, cryocoolers ...,
  - Heating, ventilation, air conditioning (HVAC) systems,
  - ...
- Due to:
  - Imbalance
  - Parts irregularity and imperfections (bearings, gears)
  - Motors (cogging, stepper motors, ...)
  - Control
  - Wear, including lubricants (cf. Rosetta)
  - ...



Source: Smet et al., AMS(2018)



Source: Bradford Space

Source: McMahon et al. ESMATS(2017)

# ... and on Ground

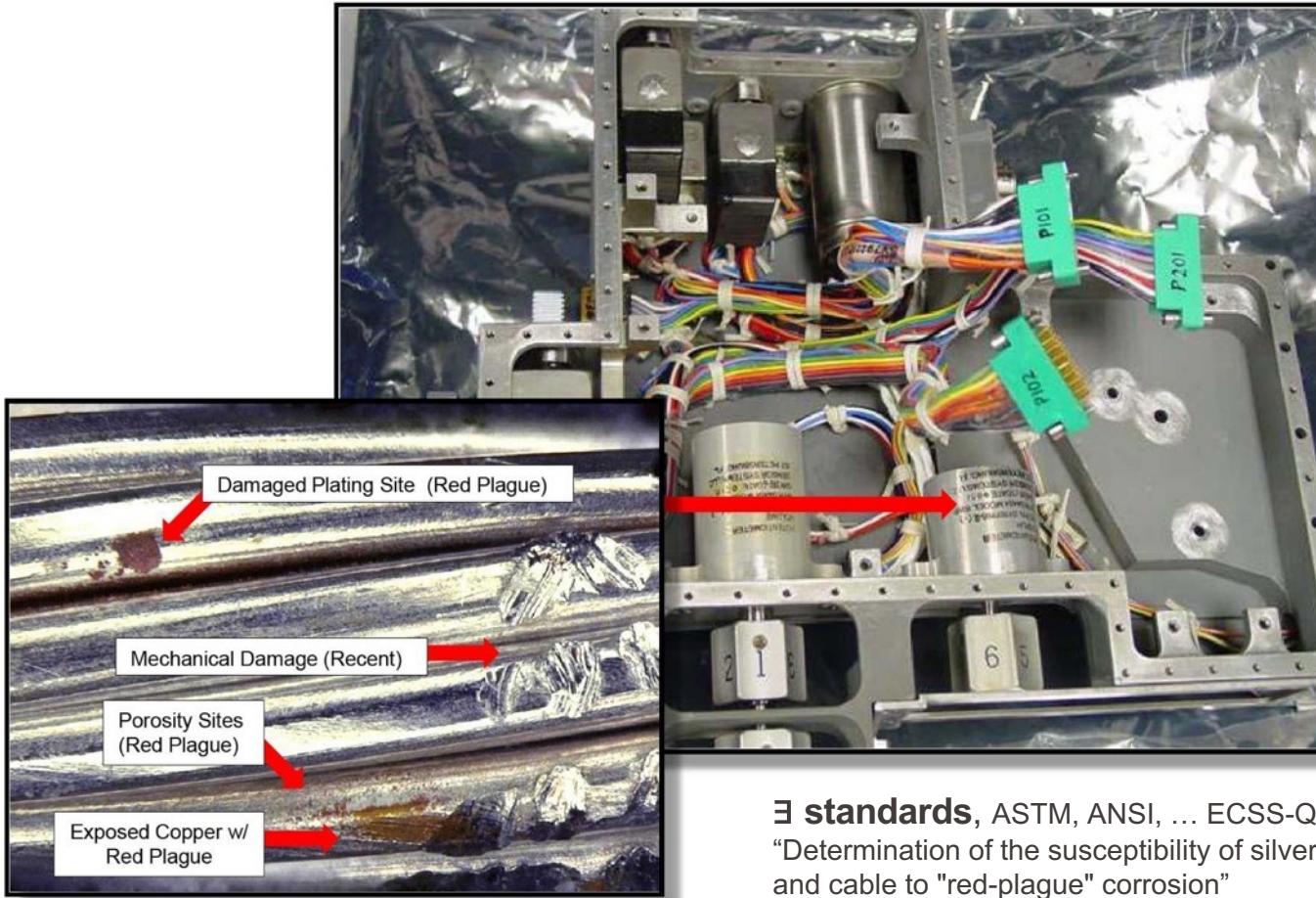
- Temperature
- Humidity
- Atmosphere
- Biological
- Transport loads
- Test



Source: Arianeespace



Source: ESA



Cannan, NACA

**Standards**, ASTM, ANSI, ... ECSS-Q-ST-70-20C  
"Determination of the susceptibility of silver-plated copper wire  
and cable to "red-plaque" corrosion"

- Space environment constrains:
  - Vacuum
    - Properties of vacuum
    - Material selection
    - Effects on design
  - Radiations
    - Ionizing, non-ionizing
    - Dependent on the mission (orbits, duration)
    - Effects on materials, components, life (material selection, shielding)
  - Vibrations and shocks
    - Sinus and random vibrations as well as shocks spectra: launch, ascent, separation, deployment (but also on ground!)
    - Micro-vibrations
- Ground environment constrains
  - Material selection and design shall also be done for MAIT